## Transportation

Federal Aviation

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act 1958) Make Model **PIPER PA-18** 1. Aircraft Serial No. Nationality and Registration Mark 18-7709117 USA - N 83679 Name (As shown on registration certificate) Address (As shown on registration certificate) 2. Owner 8224 Phyllis Place Pro-Air West, LLC San Diego CA 92123 The data identified herein complete with the carry learning and the carry learni is approved only for the above described aircraft, subject to conformity inspection person authorized in 14 CFR 43.7. Compatibility of this design change with previously approved modifications must be determined by the installer. -SAN FSDO, WP 09 Signature of FAA Inspector 4. Unit Identification 5. Type Unit Make Model Serial No. Repair Alteration (As described in item 1 above) AIRFRAME Х POWERPLANT PROPELLER Туре APPLIANCE Manufacturer 6. Conformity Statement A. Agency's Name and Address B. Kind of Agency C. Certificate No. **BOB TURNER** x□ U.S. Certified Mechanic 2634677 A&P 7237 Werner Street Foreign Certified Mechanic San Diego CA 92122-2833 ☐ Certified Repair Station ☐ | Manufacturer D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Date Signature of Authorized Individual 7. Approval for Return to Service Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is x APPROVED REJECTED FAA Flt Standards Manufacturer Other (Specify) Inspection Authorization Inspector BY **FAA Designee** Repair Station Person Approved by Transport Canada Airworthiness Group Date of Approval or Rejection Certificate or Signature of Authorized Individual Designation No. 2634677 IA

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

| 8. | Description | of | Work | Accon | nplished |
|----|-------------|----|------|-------|----------|
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(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installation of Grove Master Cylinder and Pedal Assembly on Piper Cub N 83679 (Piper PA-18 s/n 18-7709117)

This alteration replaces the original Piper brake actuators with new brake master cylinder/pedal assemblies P/N 95-1 (left) and 95-2 (right) produced by Grove Aircraft Landing Gear Systems, Inc.

Installation is accomplished by removing the original Scott Master Cylinder pedal assemblies, and mounting the Grove assemblies in the same position using the same hardware. Silicone based fluid conforming to MIL PRF 46176B or Mil 5606 fluid is used with no change to operating characteristics of the brake system. Either fluid is compatible with all operating components of the system.

Flexible brake system hoses supplied by Piper were originally automotive type, and are replaced with new stock, tested to 4000 psi and certified to have bursting pressures 5000psi and above under safety standards of 49 CFR 571.106. System pressures will be less than 650 psi.

This installation meets certification standards of CAR 3.363 and 14 CFR 23.735. All work was accomplished in accordance with standard practices of AC43.13-1B and AC43.13-2B

An operational check was performed. Performance is identical to the previously installed brake components.

Instructions for Continued Airworthiness consist of checking the brake system for condition and leaks in accordance with 14CFR Part 43 Appendix D at each annual inspection and checking the brake fluid level each 50 hours.

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Additional Sheets Are Attached